



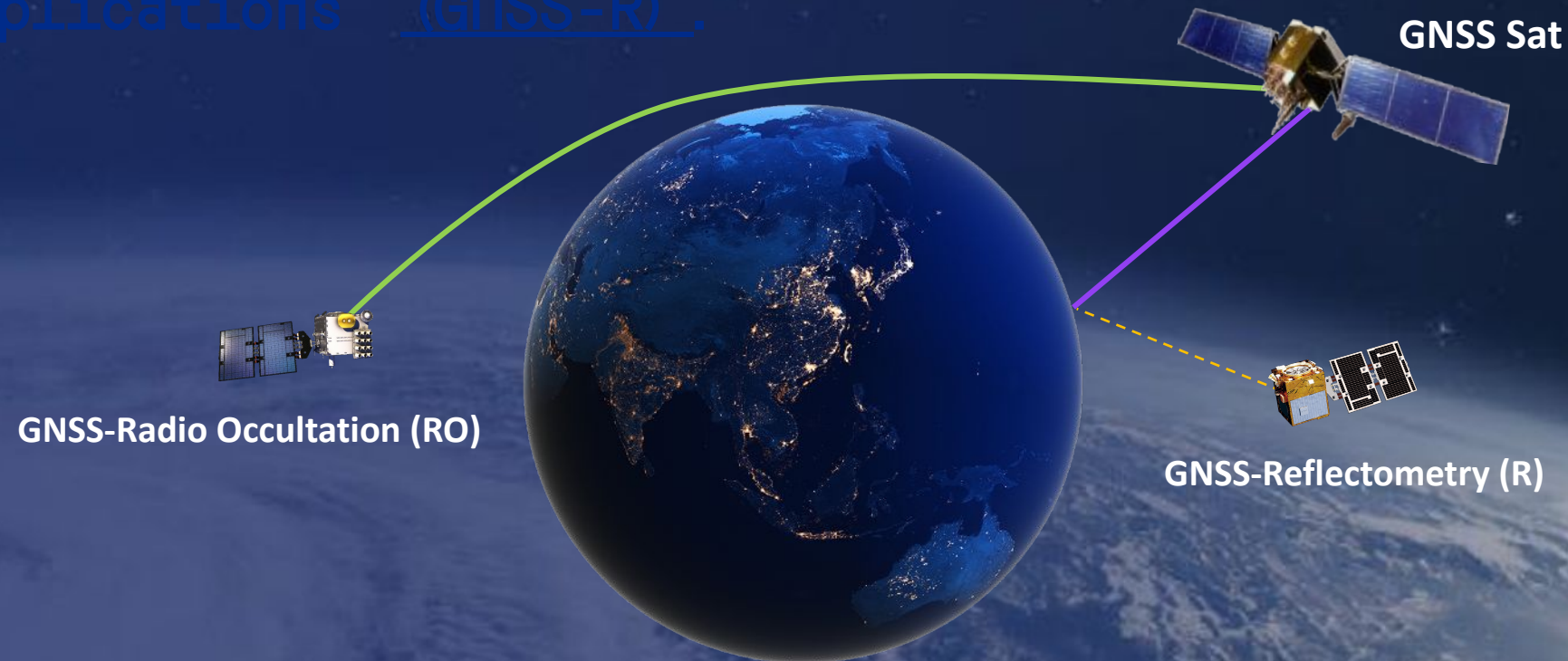
# Status Update for Triton GNSS-R Mission

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Taiwan Space Agency



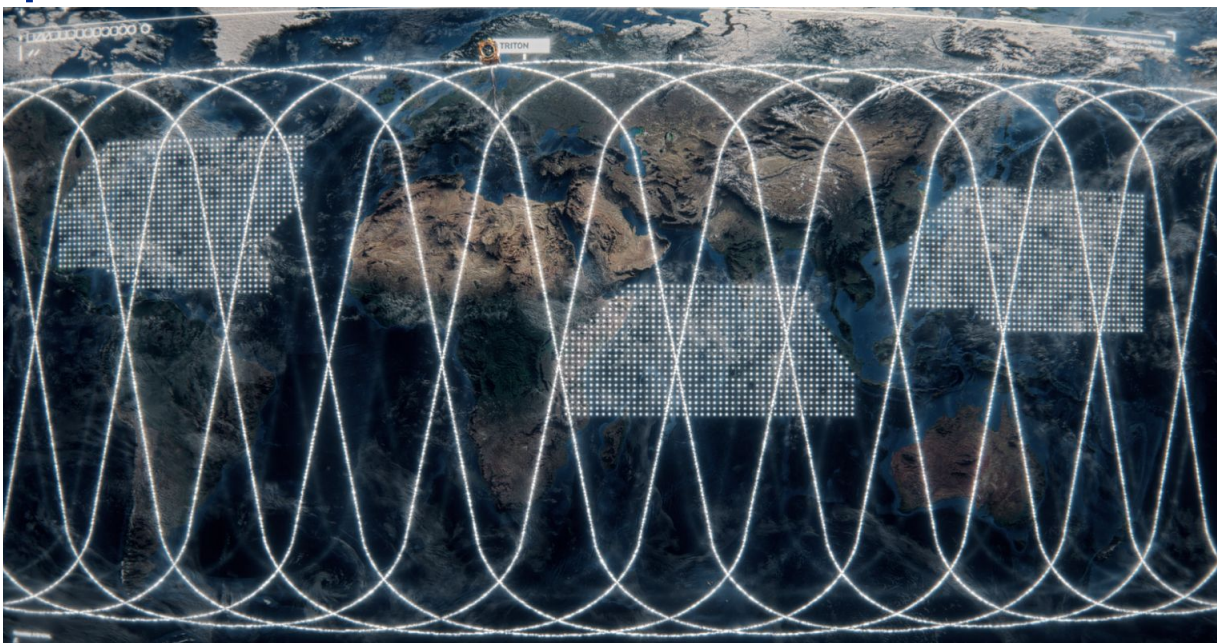
獵風者衛星  
TRITON

- FORMOSAT-7 constellation demonstrated the promising application of GNSS-RO for weather prediction.
- Triton was once one of FORMOSAT-7/COSMIC -2 constellation satellites.
- Triton is now providing Earth surface reflected signals from GNSS satellites for remote sensing applications (GNSS-R).



# Triton Mission

- Triton processes GNSS reflected signals and retrieve them to be wind speed for weather prediction (GNSS-R).
- To flight demonstrate Taiwan built components and technologies. If proven viable, could boost Taiwan's

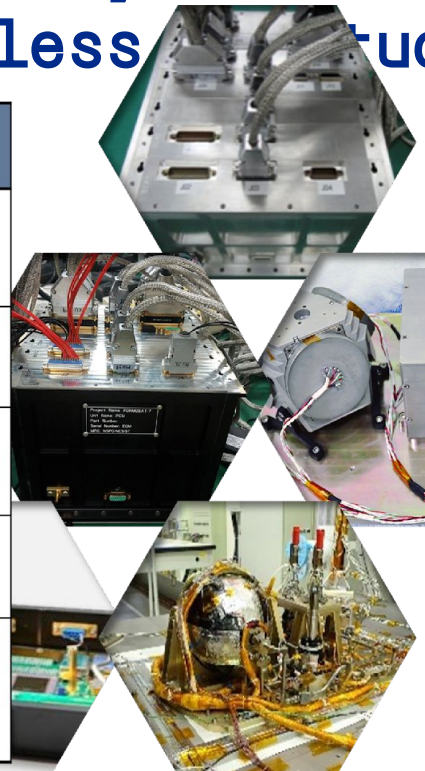


GNSS-R High Gain Ant

# Taiwan Built Components and Technologies

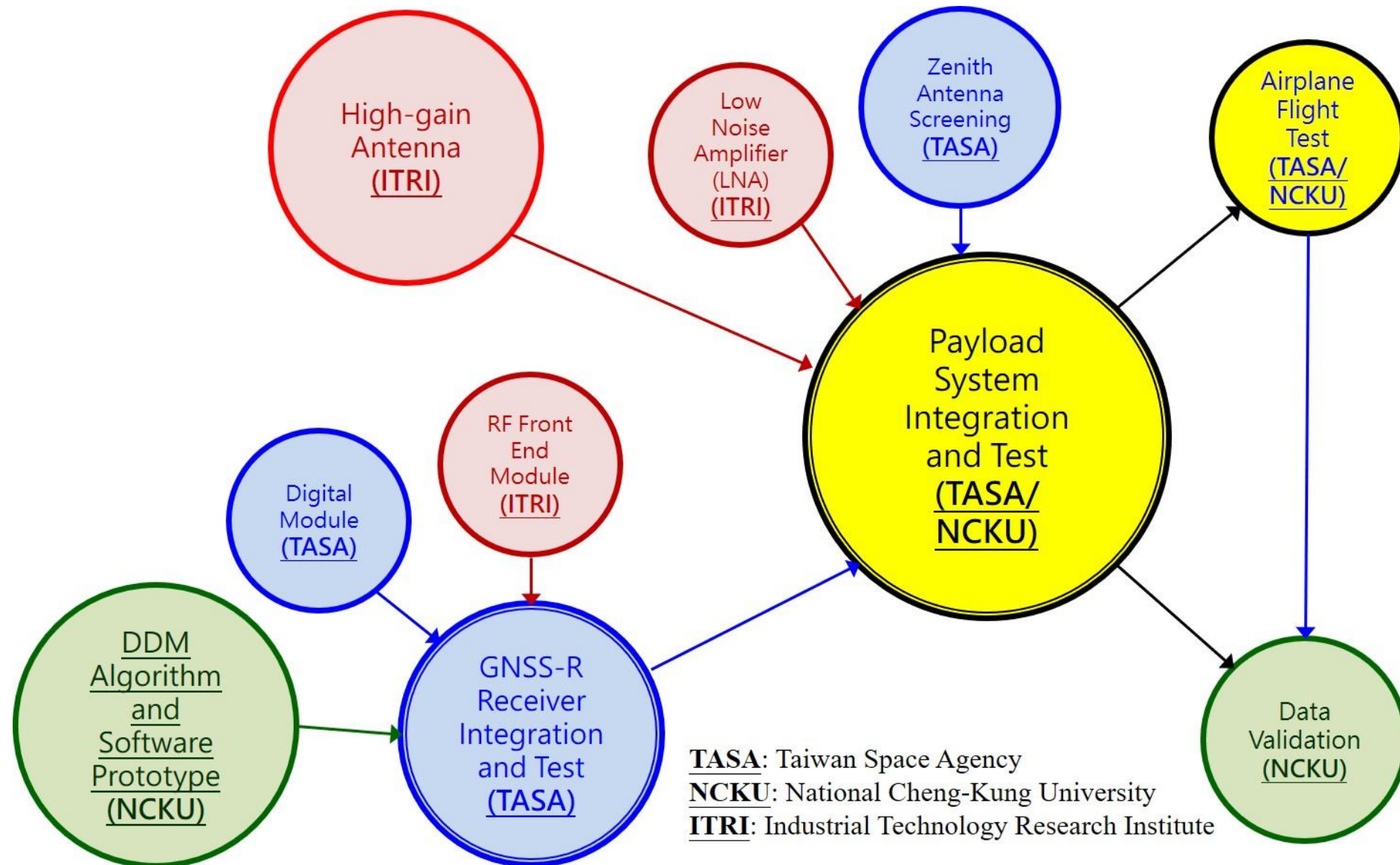
- GNSS reflectometry science payload (GNSS-R)
- Second generation of On Board Computer (OBC) and Power Control Unit (PCU)
- Fiber Optical Gyro (FOG) and GPS receiver (GPSR)
- Solar array peak-power tracking (PPT) controller and micro-stepping solar array driver
- Gyro/Stellar and gyroless attitude determination algorithm
- Data

Key Component	Validation Result
On Board Computer	Passed
Power Control Unit	Passed
Fiber Optic Gyro	Passed
GPS Receiver	Passed
Propulsion Demonstration Module (H2O2)	Passed



Key Technology	Validation Results
Micro-stepping solar array controller	Passed
Solar array peak power tracking controller	Passed
Gyro-less attitude determination system	Passed
Gyro/Stellar Attitude Determination System	Passed
Data Relay System	Passed

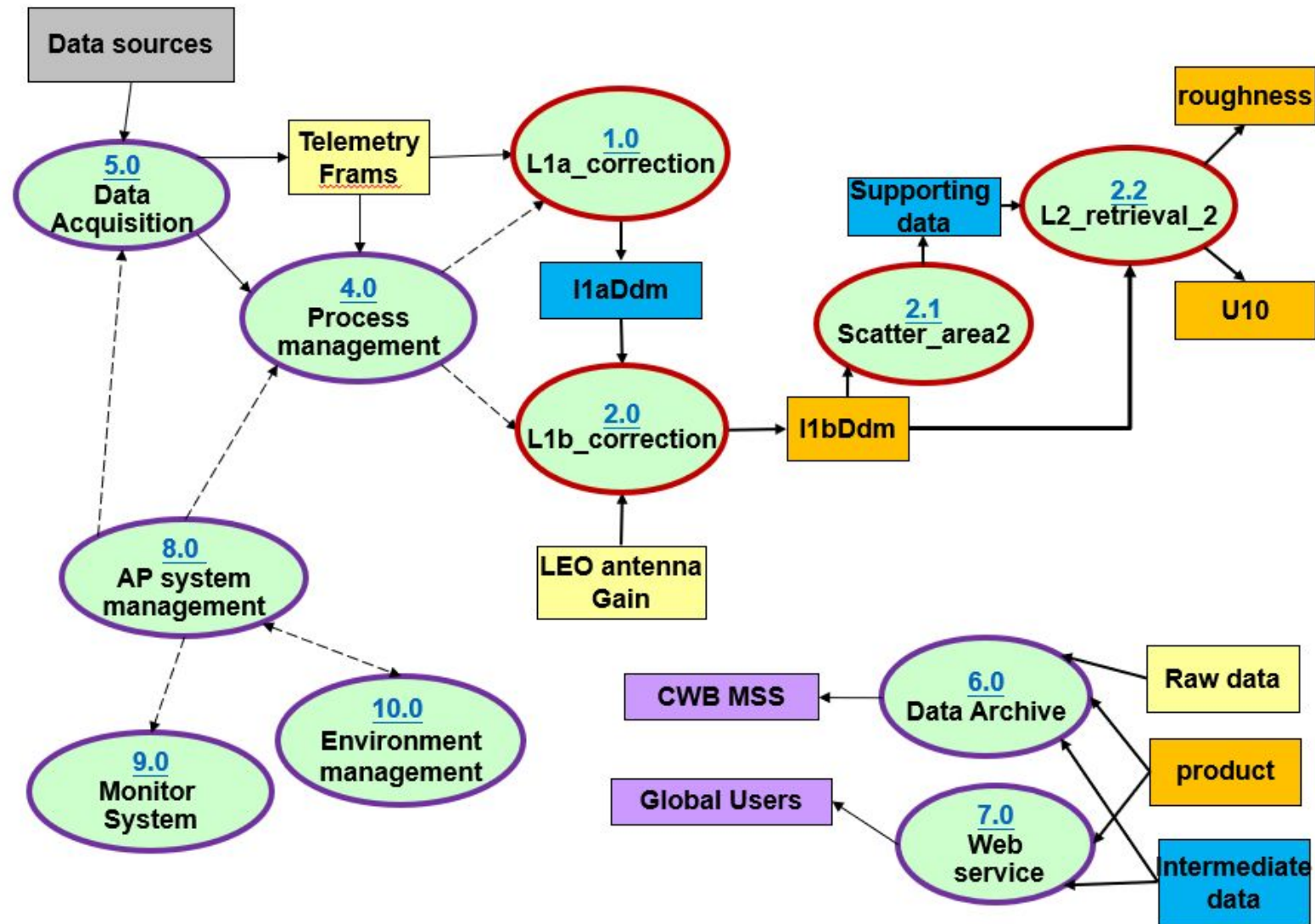
# GNSS-R Payload Development Approach



# Satellite Status @ LEOP

- Sun pointing was achieved less than 2 orbit, very soon.
- Satellite enter to Normal Mode (NM) in 2 days, then provides full performance.
- GNSS-R payload was switched on 3<sup>rd</sup> day after launch.
- Calibration burn was performed in the 2<sup>nd</sup> week after launch, using P2H4 propulsion module.
- One Automatic Reconfiguration Order (ARO) was occurred due to single event upset of On-Board-Computer.
- Few auto switch off of GPSR and GNSS-R were occurred due to single event upset. After power on, the functionalities were recovered.
- GNSS-R payload data calibration plan:
  - ✓ Payload parameter decision: 1 month after launch.

# GNSS-R Retrieval Process



# DDM Calibration

$C$  : the DDM values in counts

$G$  : the total instrument gain applied to the incoming signal and noise in counts per watt

$P_g$  : the scattered signal power received by the instrument in watts

$P_n$  : the noise power generated by the received signal with wrong PRN

$P_a$  : the thermal noise power received by the antenna in watts

$$P_a = k * T_a * B_w$$

$P_r$  : the thermal noise power generated by the instrument in watts

$$P_r = k * T_r * B_w$$

$k$  : Boltzmann's constant =  $1.380649 \times 10^{-23}$

$T_a$  &  $T_r$  : antenna and receiver temperature

$B_w$  : signal bandwidth

The value of each bin in DDM can be described by

$$C = G(P_g + P_n + P_a + P_r)$$

The noise floor  $C_n$  of DDM can be described by

$$C_n = G(P_n + P_a + P_r)$$

due to the  $P_n$  of each DDM is different, it needs to be removed to calculate  $G$  by

$$C_{min} = G(P_a + P_r)$$

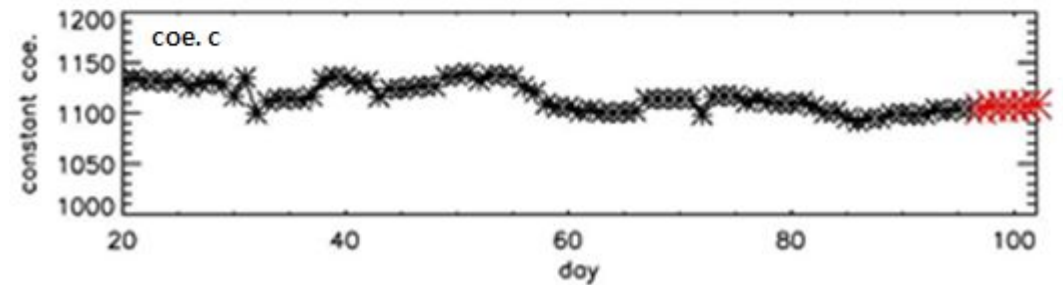
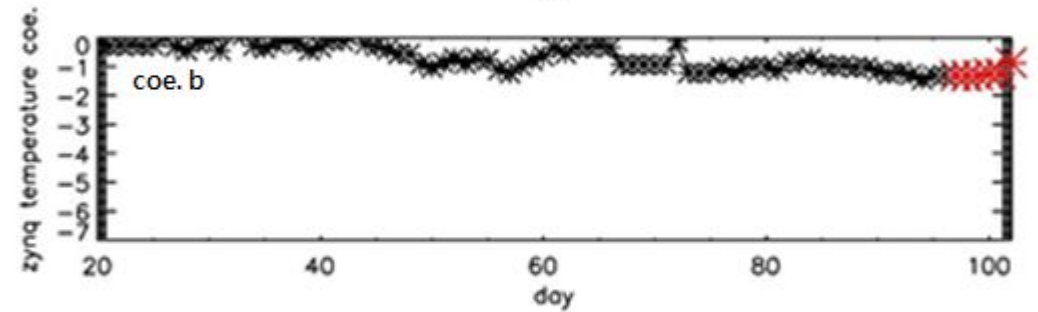
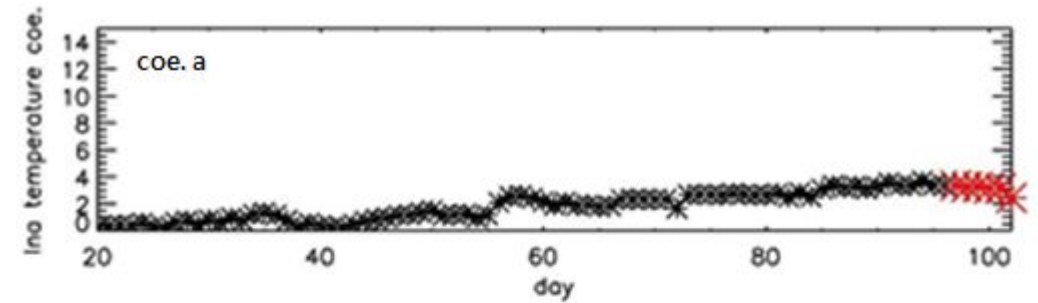
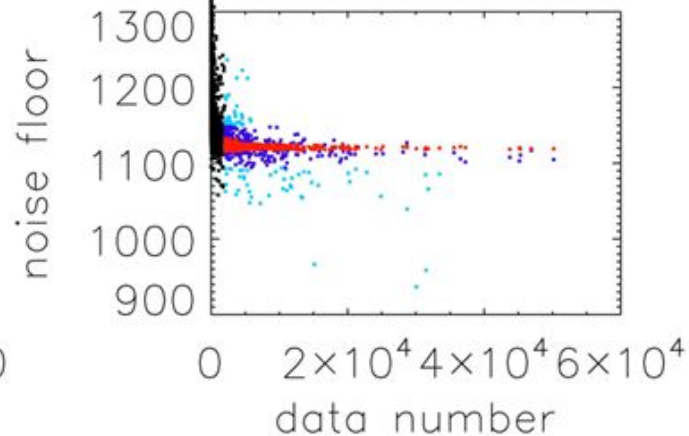
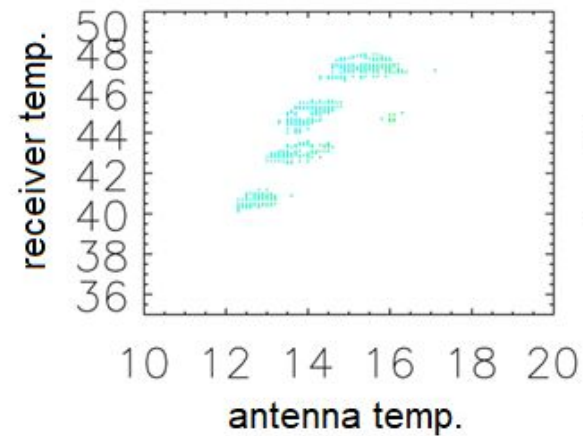
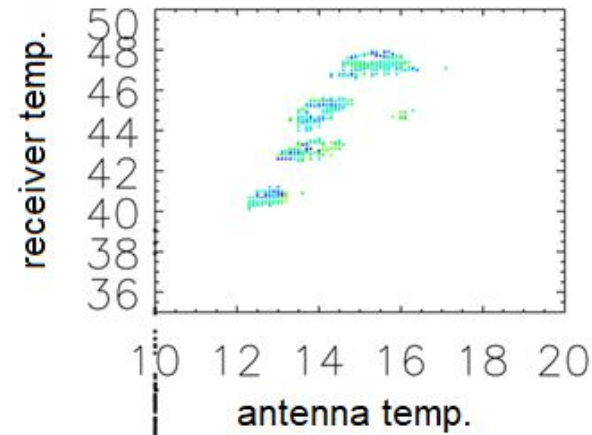
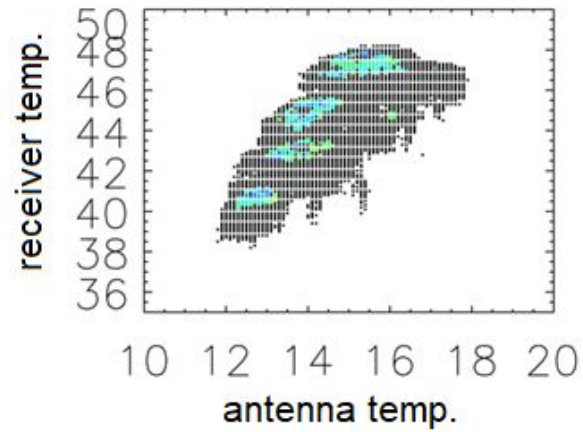
then  $P_g$  can be derived by

$$P_g = (C - C_n) \frac{(P_a + P_r)}{C_{min}}$$

$$(noise\ floor, C_{min}) = a \times (antenna\ temperature) + b \times (receiver\ temperature) + c$$

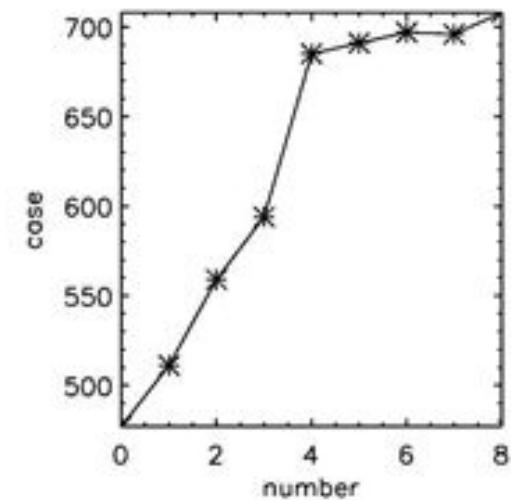
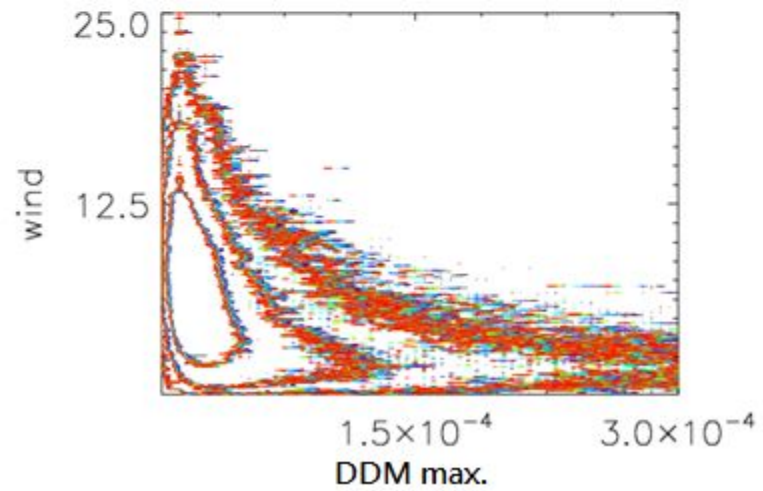
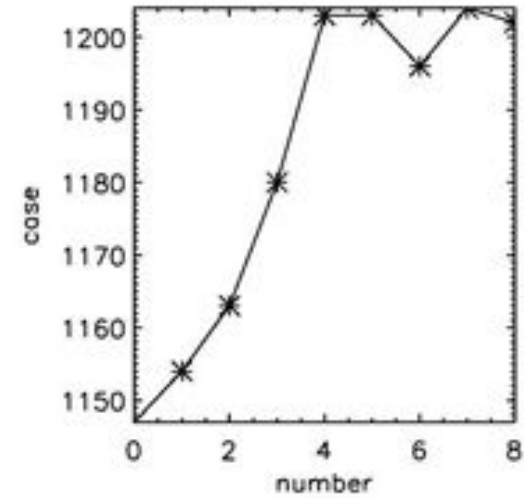
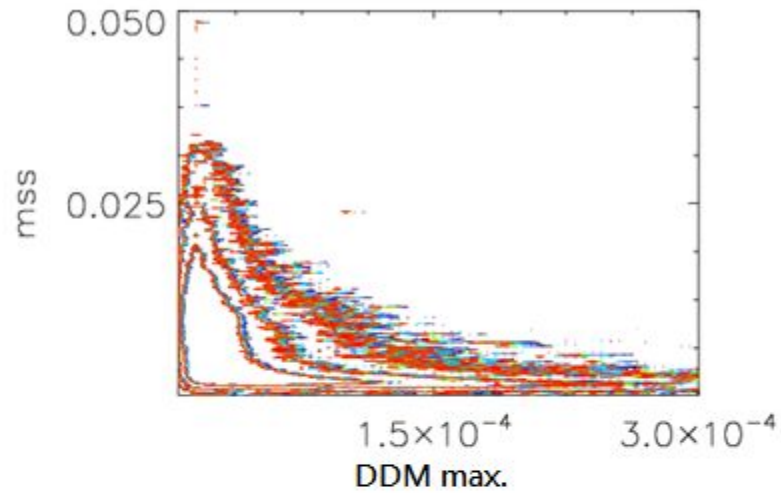


# DDM Calibration



$$(noise\ floor, C_{min}) = a \times (antenna\ temperature) + b \times (receiver\ temperature) + c$$

# DDM Calibration

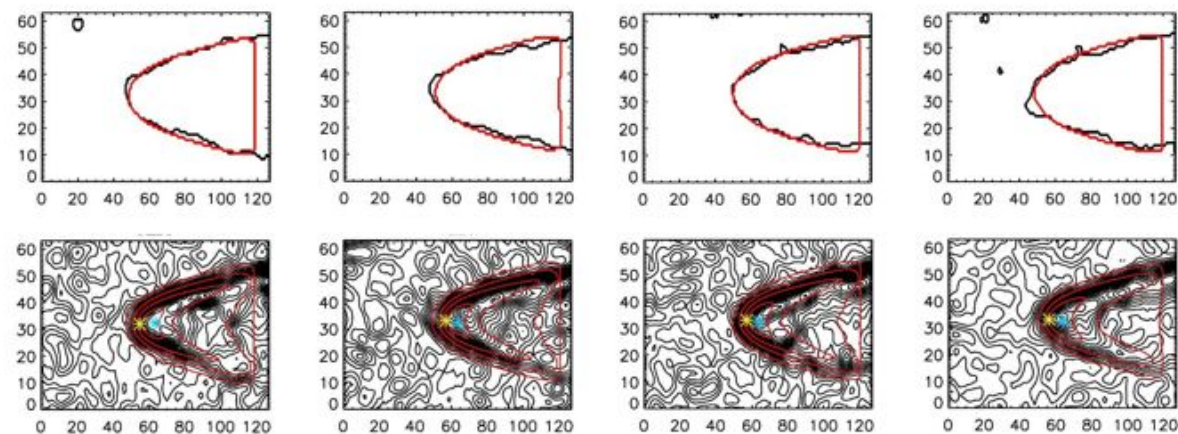
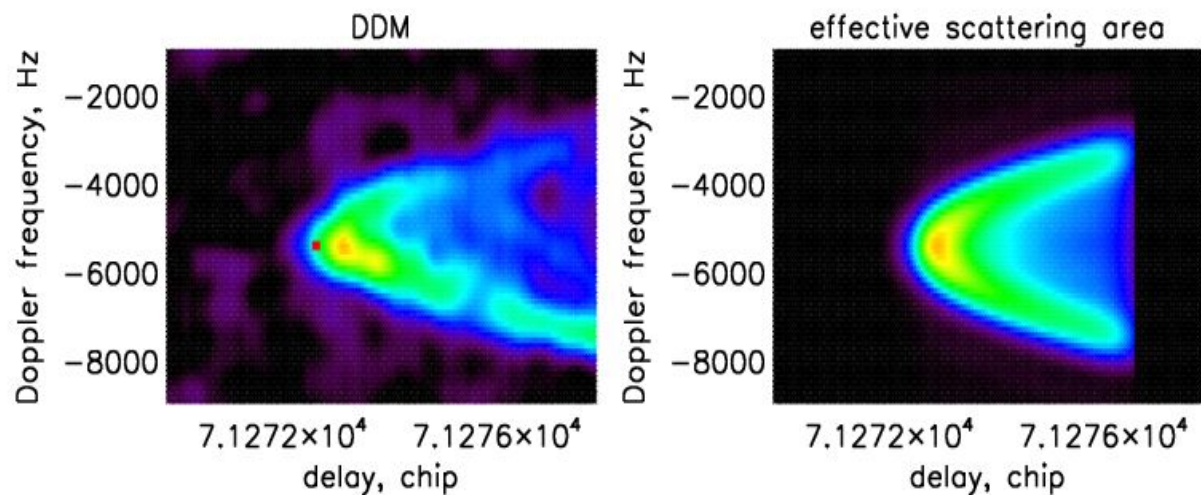


# Specular Point Determination

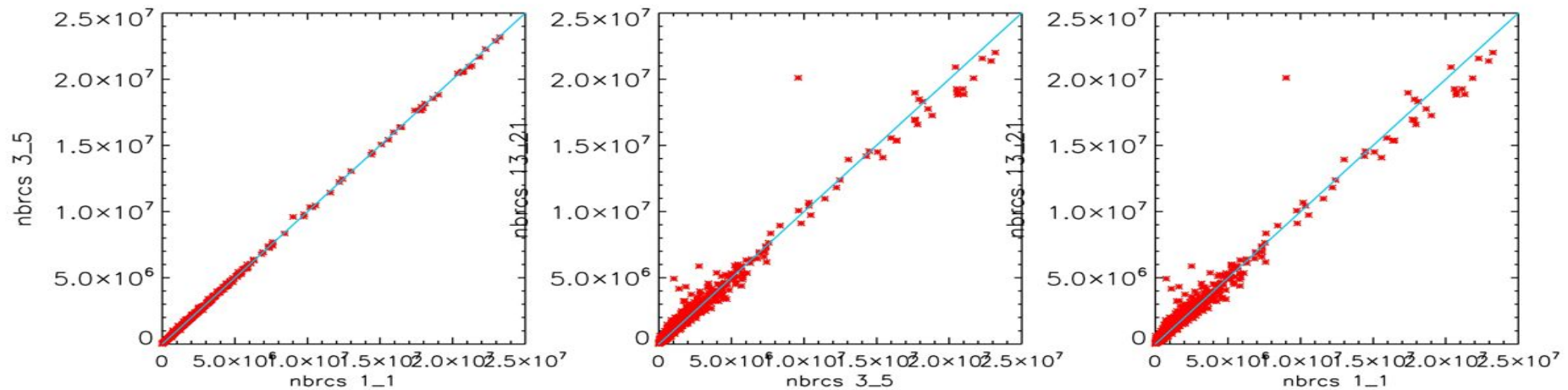
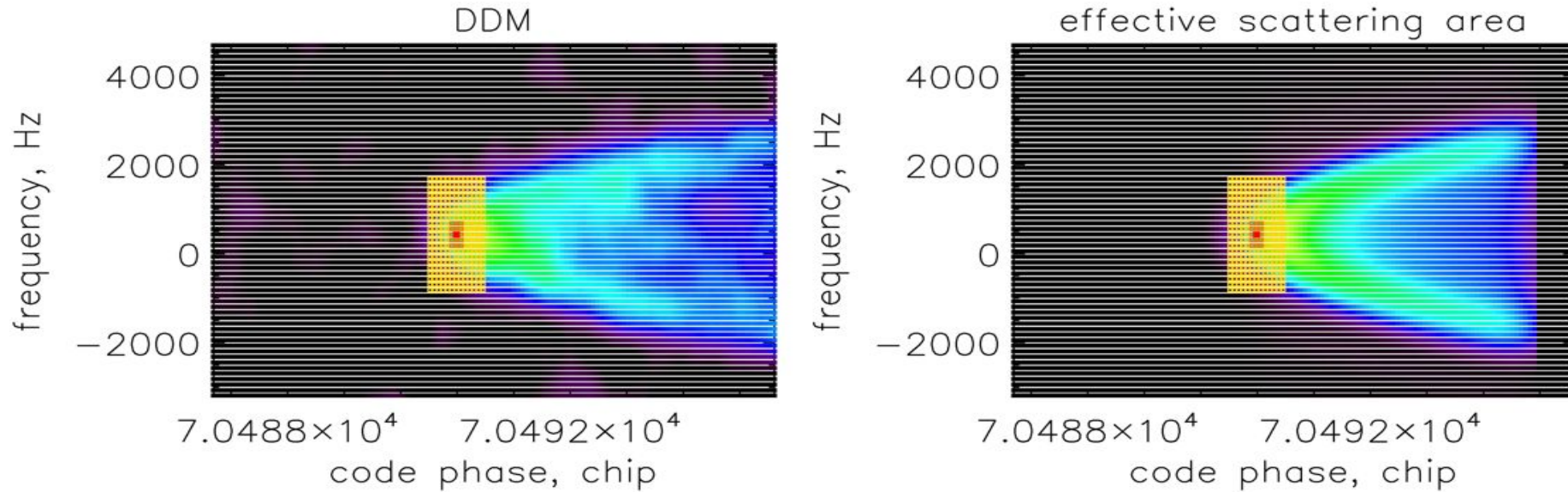
$$\langle P(\hat{t}, \hat{f}^D) \rangle = \frac{P^T \lambda^2}{(4\pi)^3} \iint_S \frac{G^T G^R}{R_0^2 R^2} \Lambda^2 \left( \frac{\delta \tau}{\tau_c} \right) S^2 \left( \frac{\delta f^D}{T_i} \right) \sigma_0 d^2 r$$

$$\langle P(\hat{t}, \hat{f}^D) \rangle = \frac{P^T \lambda^2 G^T G^R \sigma_0 \bar{A}_{eff}}{(4\pi)^3 R_0^2 R^2}$$

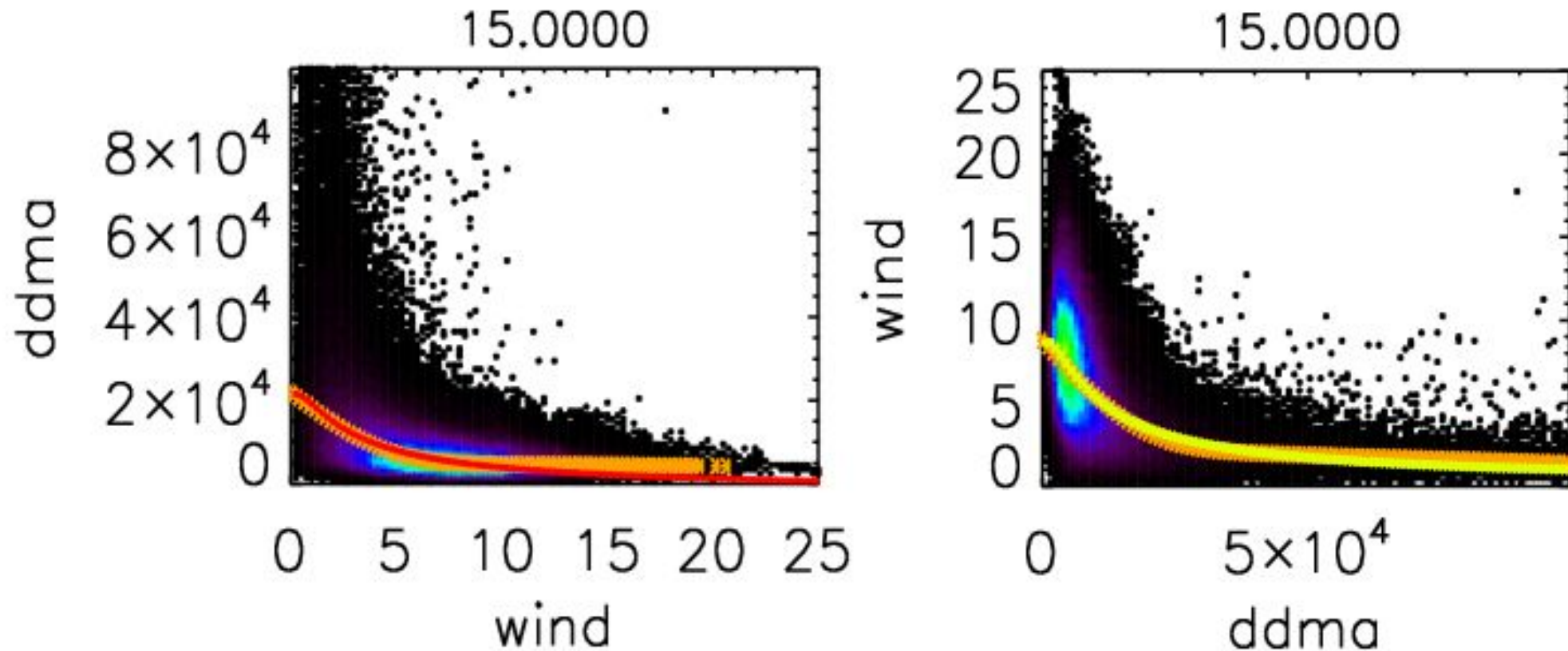
$$P^T G^T \sigma_0 = \langle P(\hat{t}, \hat{f}^D) \rangle \frac{(4\pi)^3 R_0^2 R^2}{\lambda^2 G^R \bar{A}_{eff}}$$



# Normalized Bistatic Radar Cross Section (NBRCs)

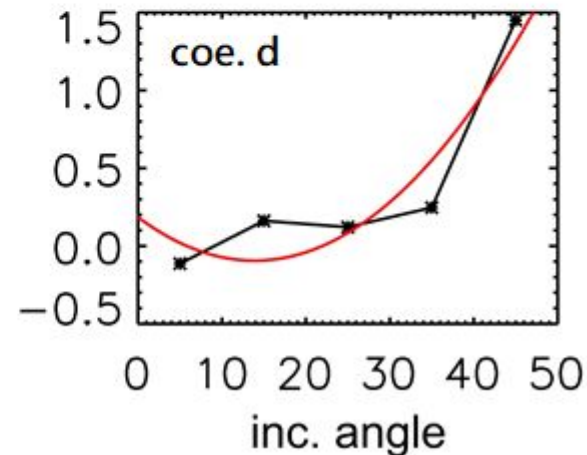
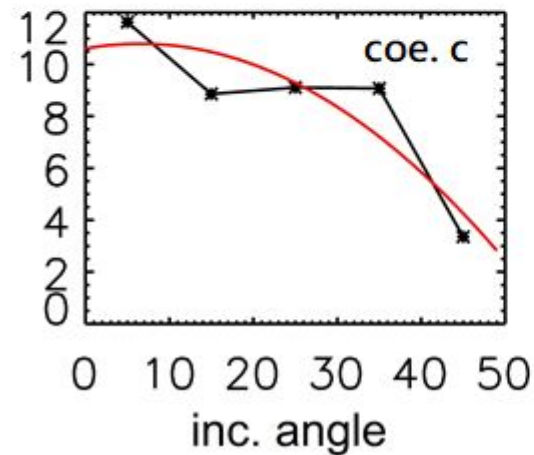
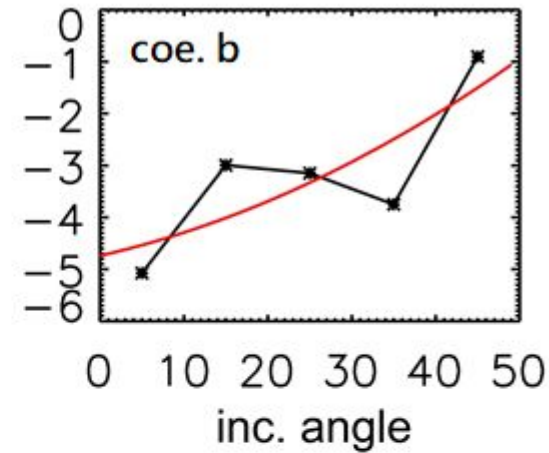
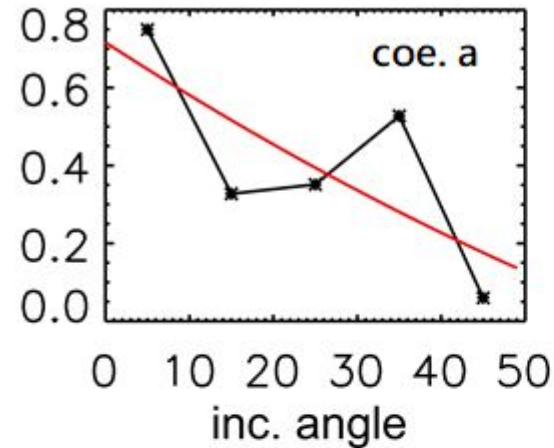


# Geophysical Model Function (GMF)



$$y = ax^{-3} + bx^{-2} + cx^{-1} + d$$

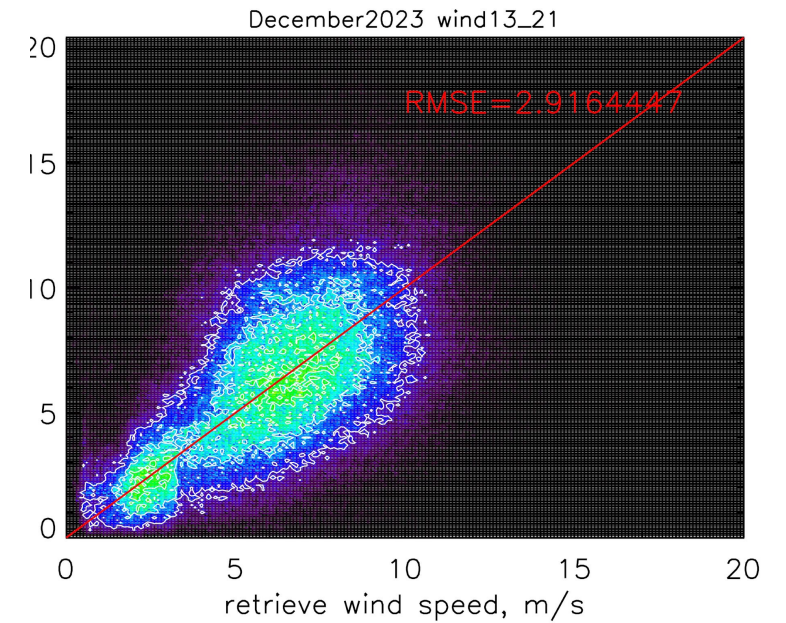
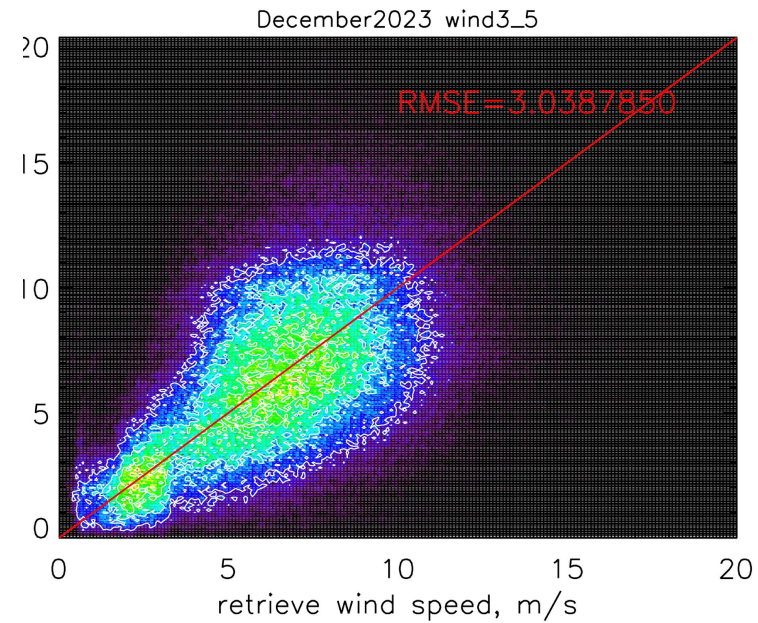
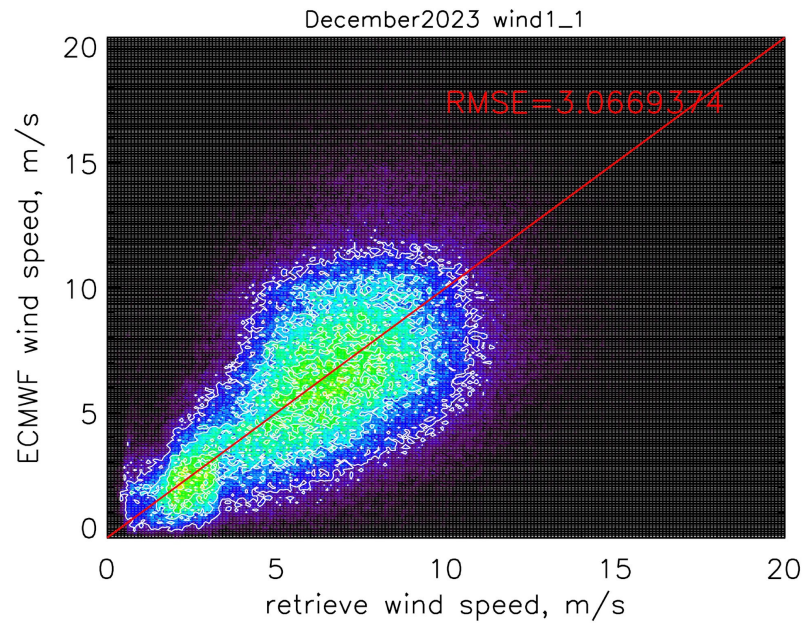
# Coefficients of GMF



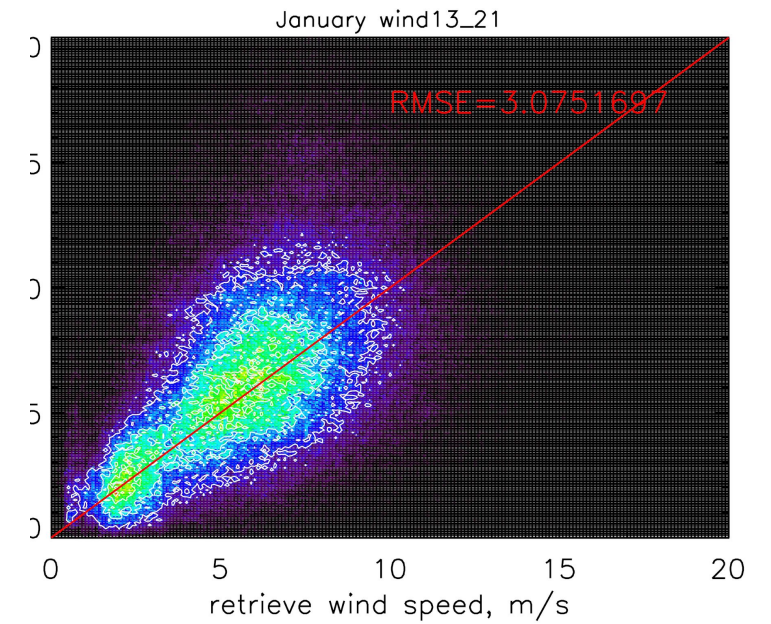
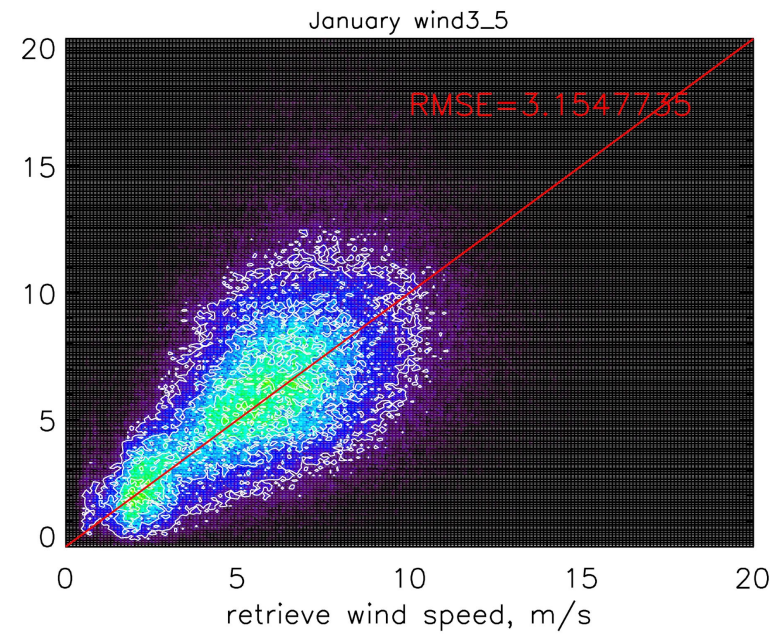
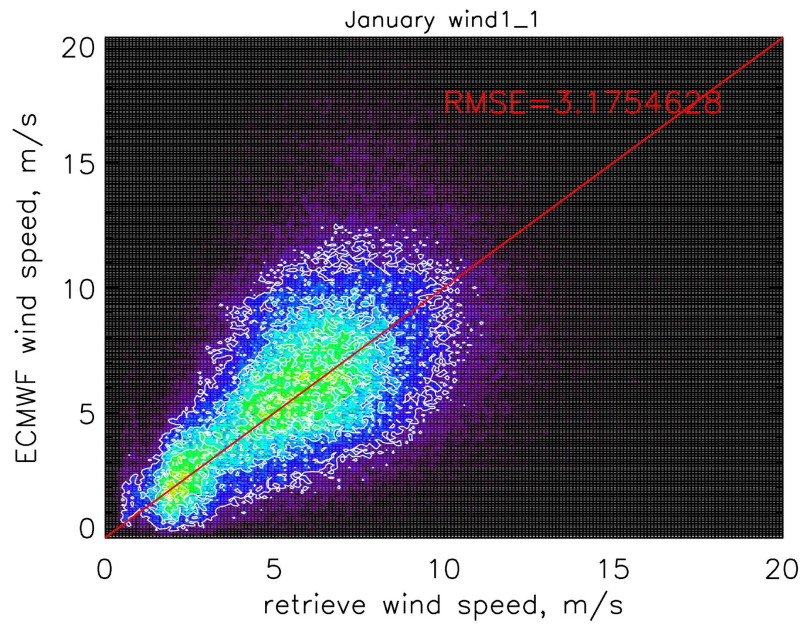
$$y = ax^{-3} + bx^{-2} + cx^{-1} + d$$

$$coe_{inc} = P_{coe}(inc)^2 + Q_{coe}(inc) + R_{coe}$$

# Wind Speed Retrieval Results (Dec. 2023)

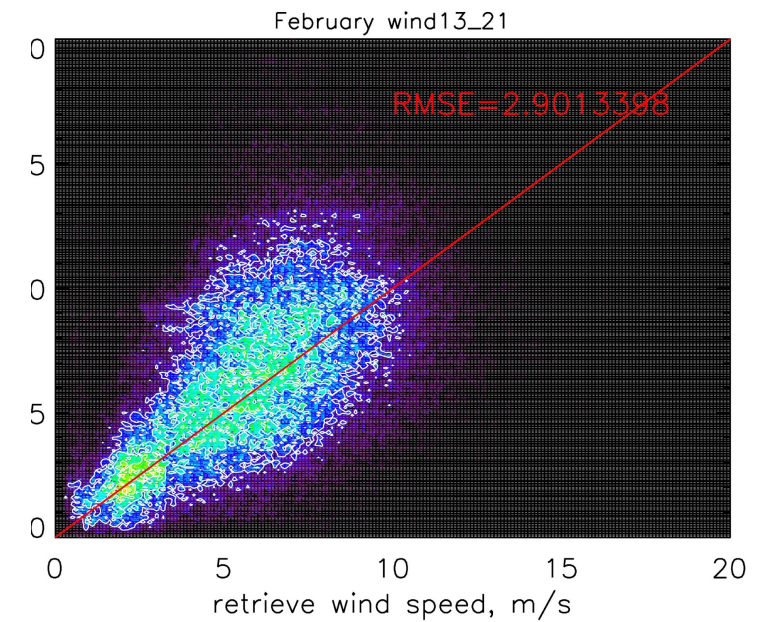
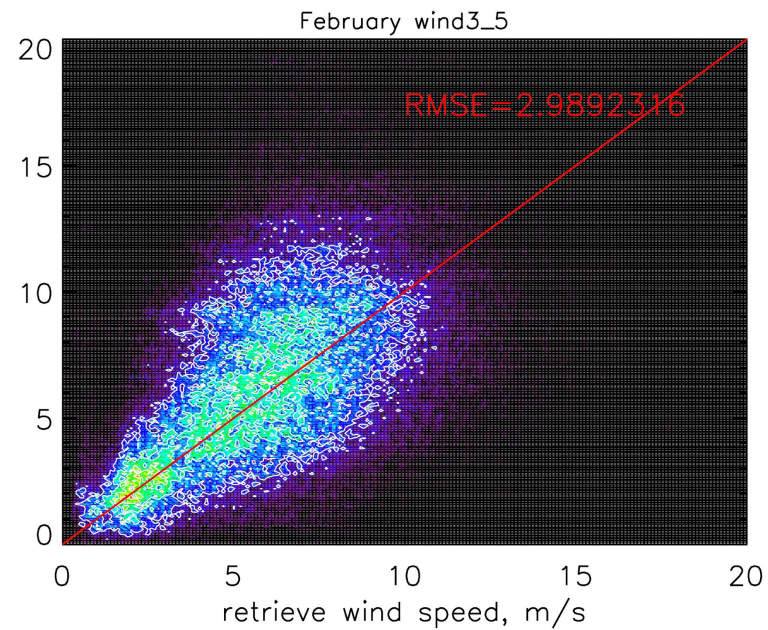
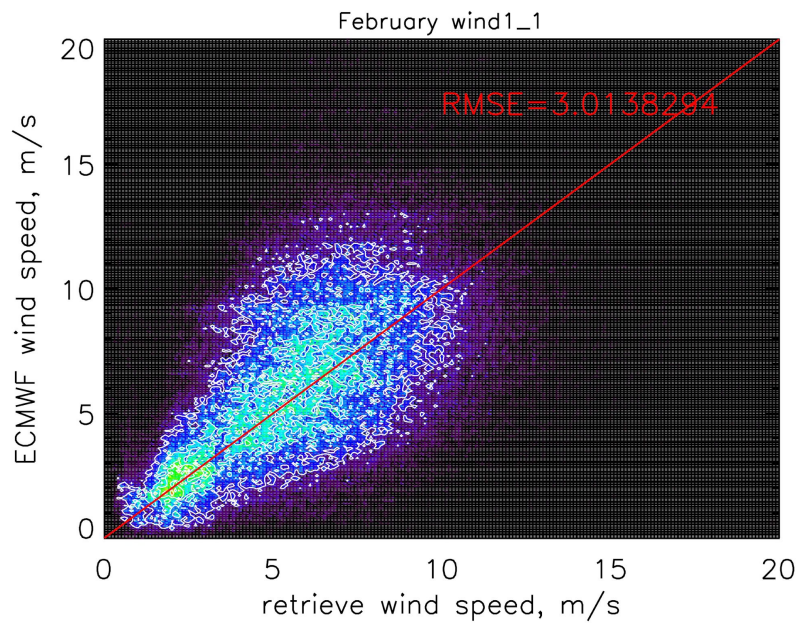


# Wind Speed Retrieval Results (Jan. 2024)

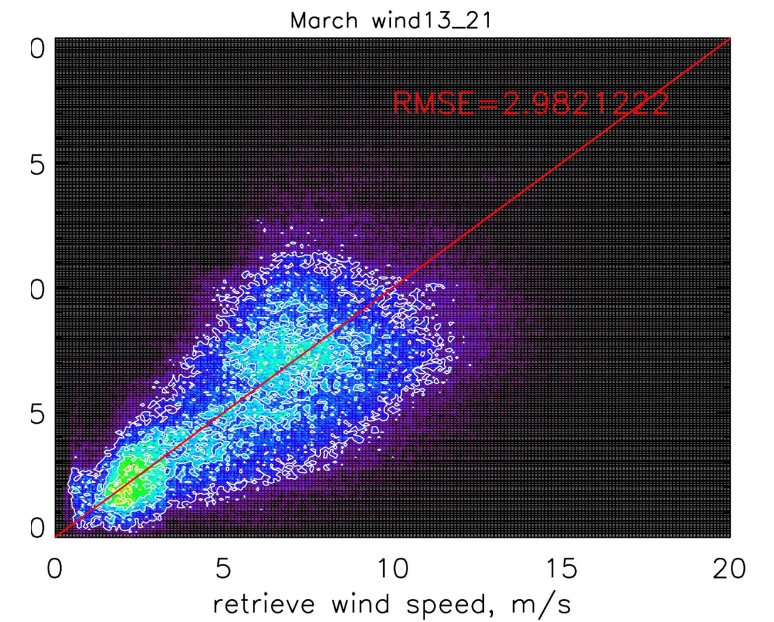
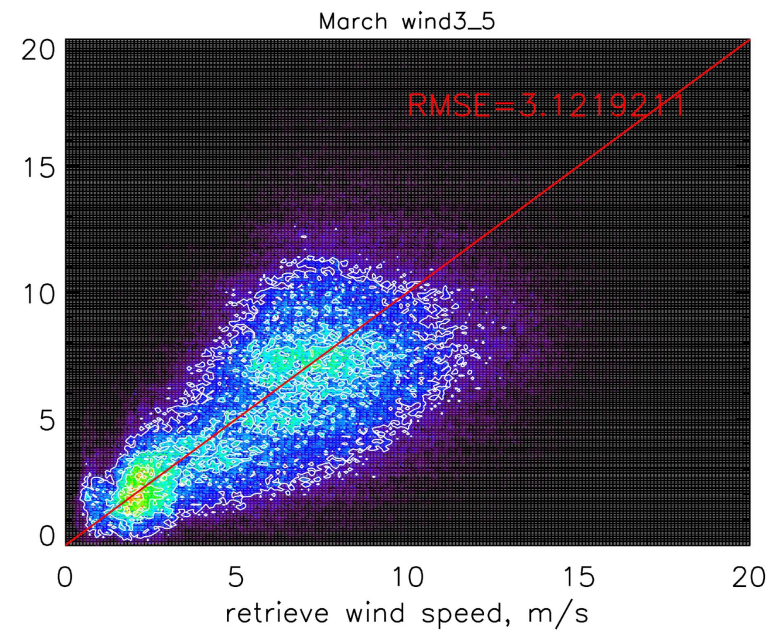
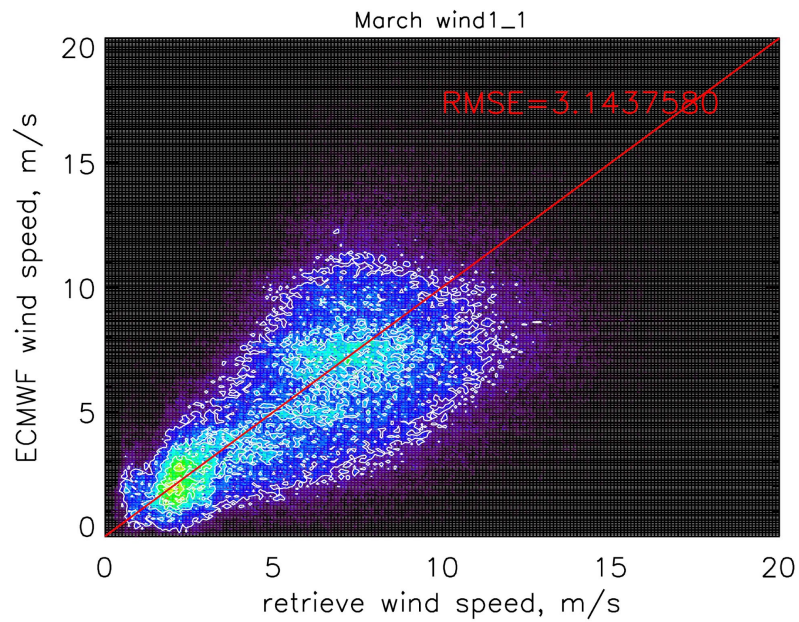




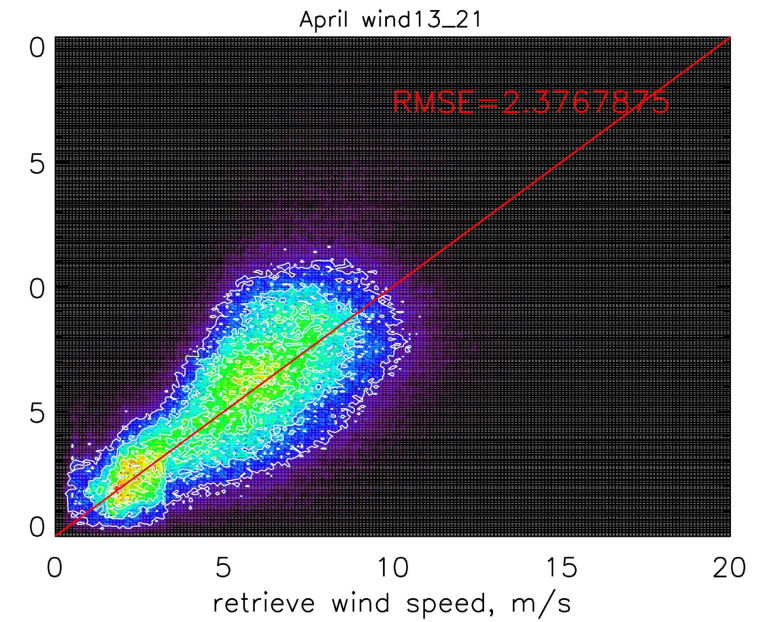
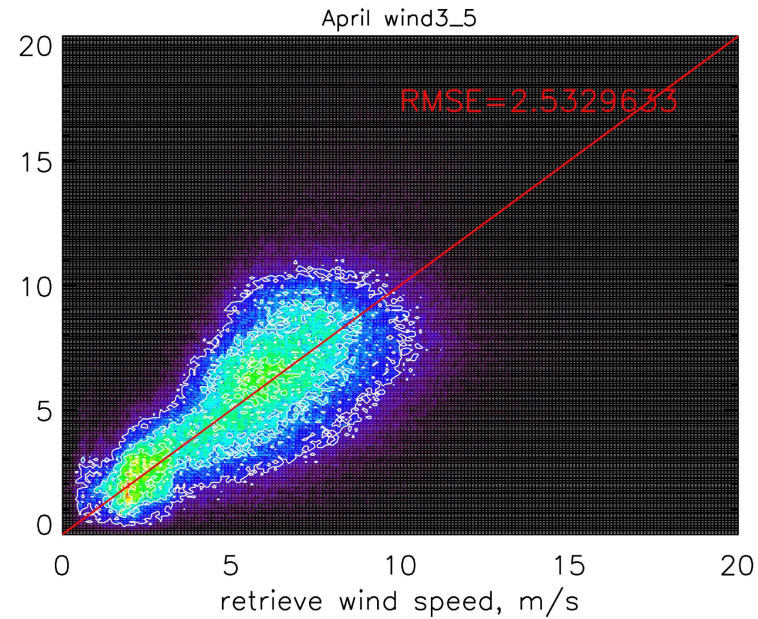
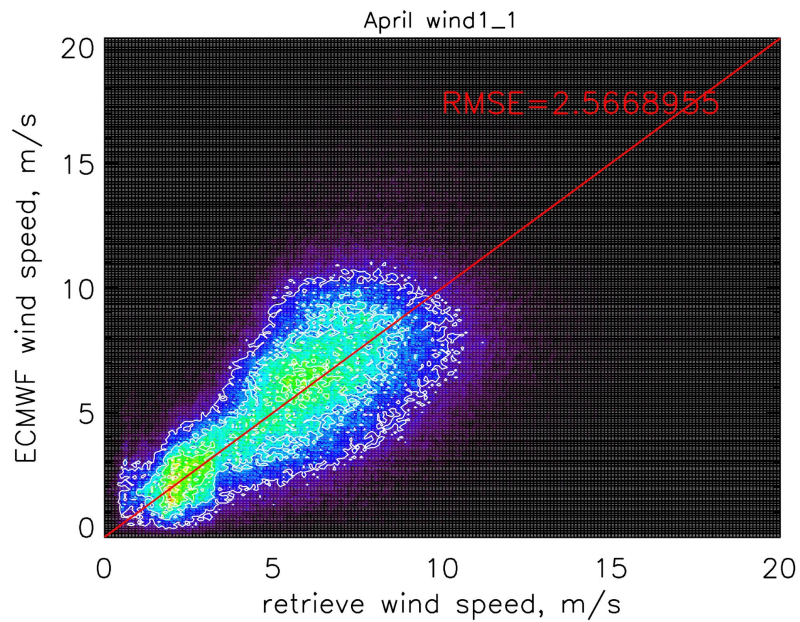
# Wind Speed Retrieval Results (Feb. 2024)



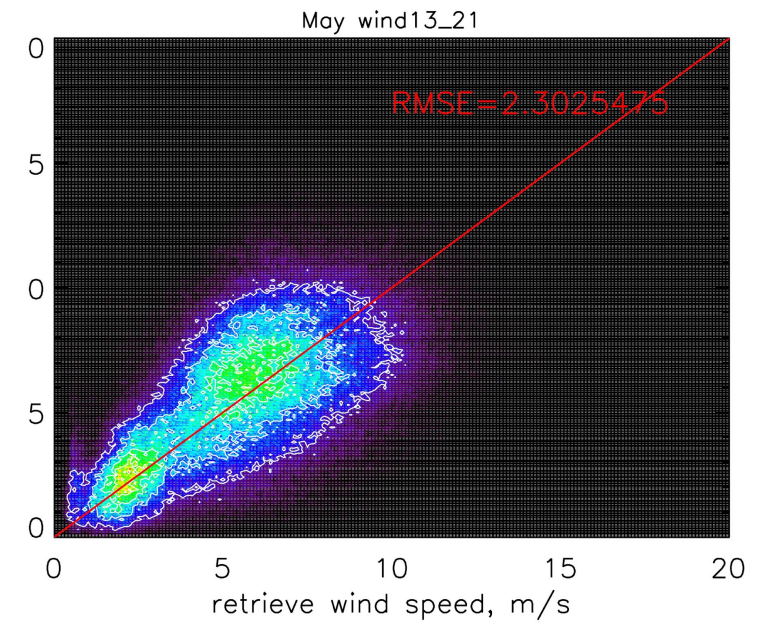
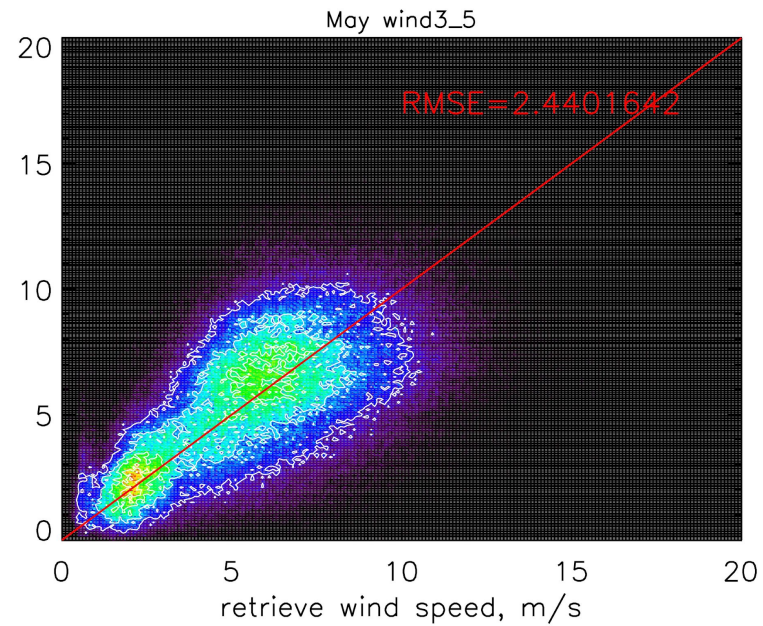
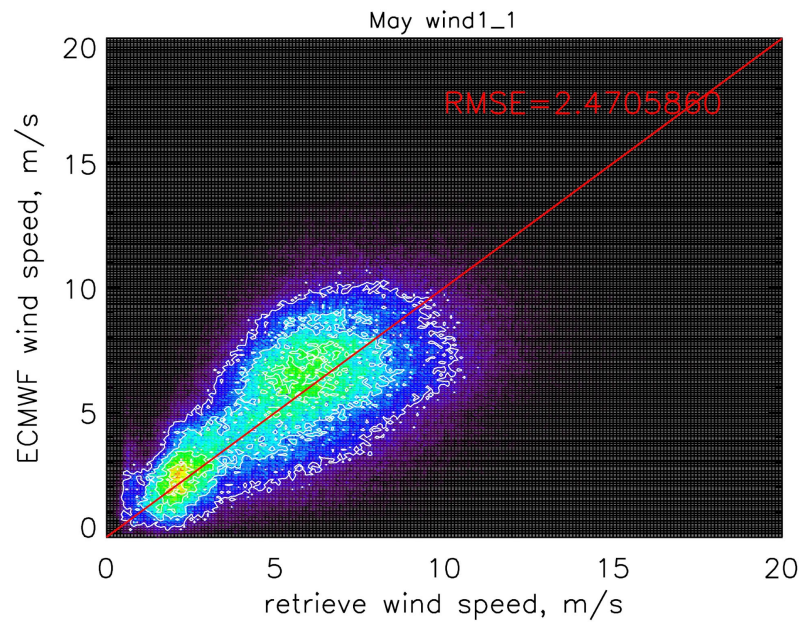
# Wind Speed Retrieval Results (Mar. 2024)



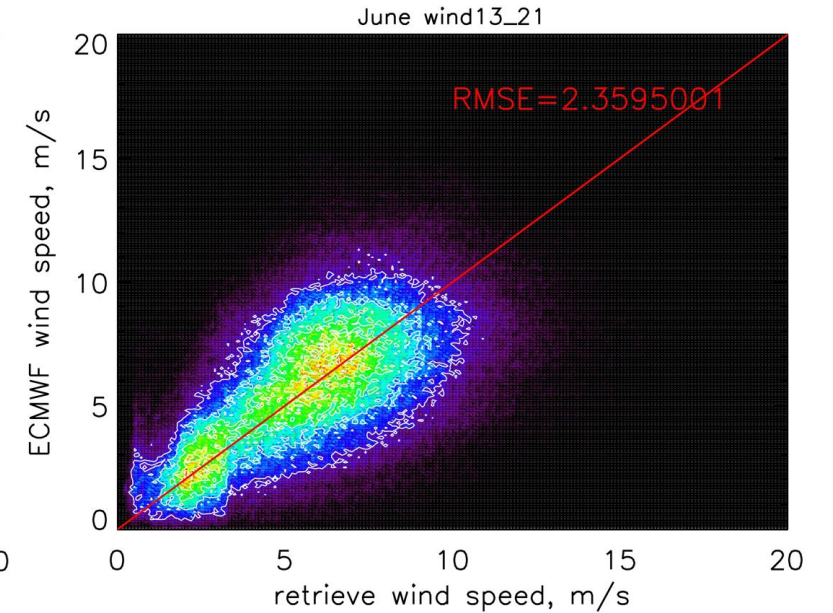
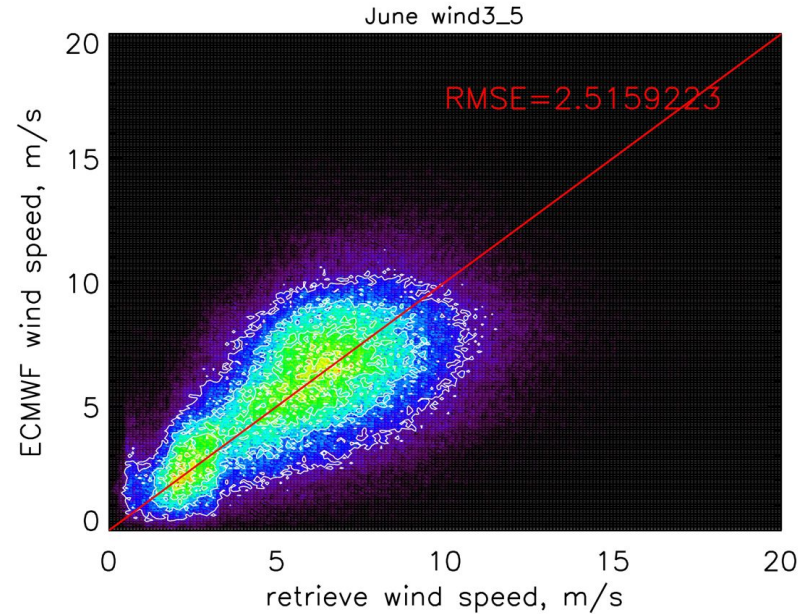
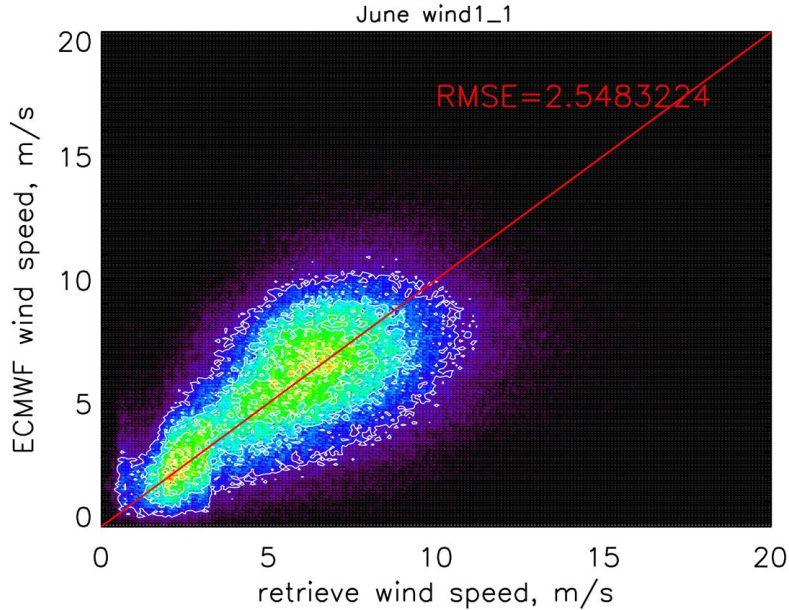
# Wind Speed Retrieval Results (Apr. 2024)



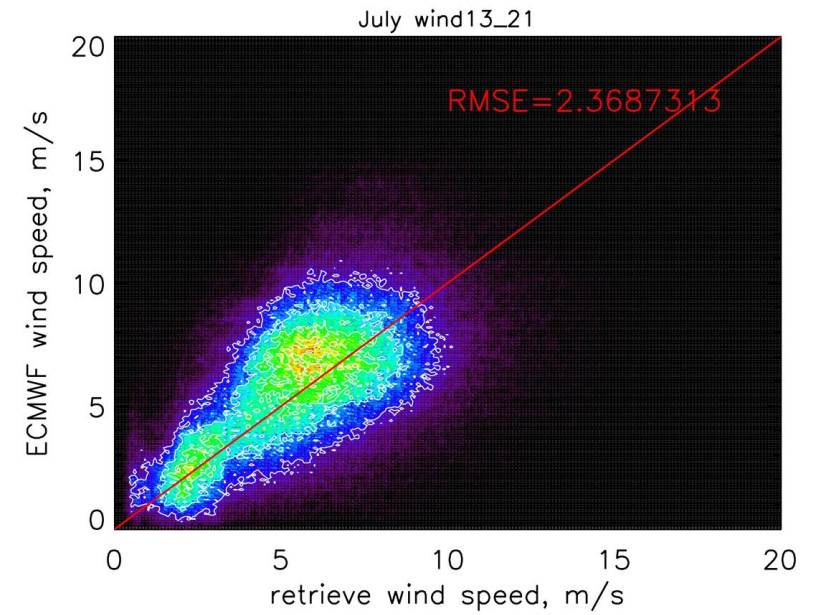
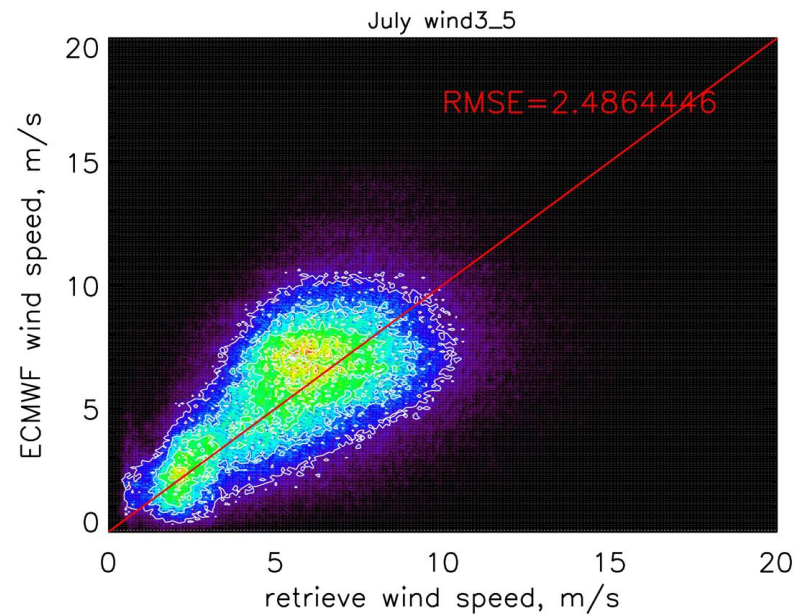
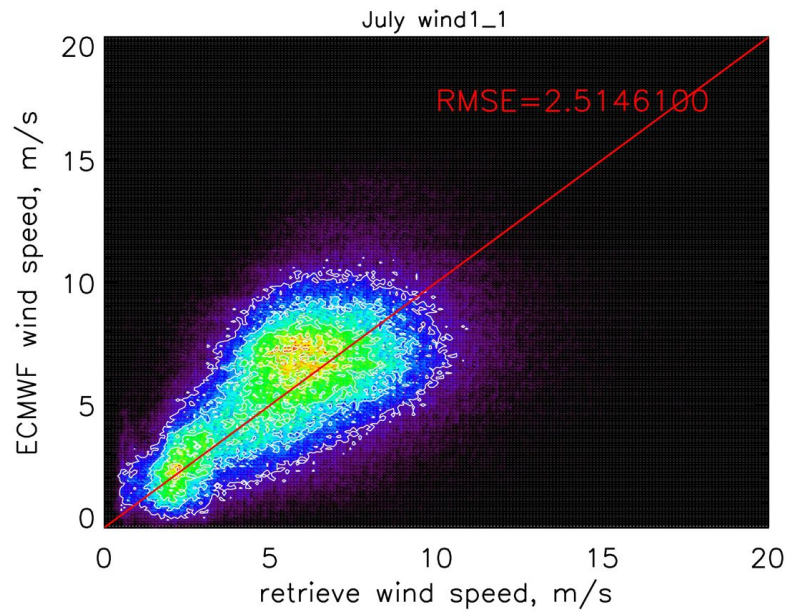
# Wind Speed Retrieval Results (May 2024)



# Wind Speed Retrieval Results (Jun. 2024)



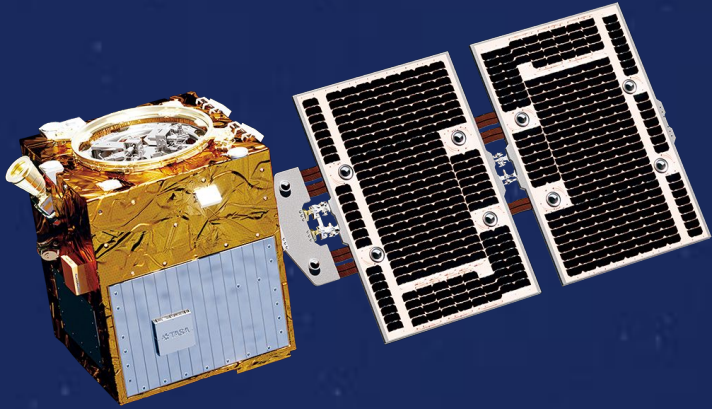
# Wind Speed Retrieval Results (Jul. 2024)



# Conclusions

- ◆ The RMSE of retrieval ocean surface wind speed compared with ECMWF is around 2.7 m/s.
- ◆ The RMSE of retrieval ocean surface wind speed before Mar. 2024 is around 3 m/s and around 2.4 m/s after Mar. 2024.
- ◆ Prepare TRITON products v2.0 (not later than mid-2025)
  - DDM calibration with ΠBRCS then rebuild GMF
  - Minimum variance method to combine two GMF wind speed





*Thanks for Your  
Attention*